

Title: Strong Gravity Lecture

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Collection: Strong Gravity 2023/24

Date: April 04, 2024 - 10:15 AM

URL: <https://pirsa.org/24040011>

In ergoregion u^r

$$g_{tt} > 0$$

$$u^\phi > 0$$

As $r \rightarrow r_+$

$$\Omega = \bar{\omega} / 2r_+$$

$$-1 = \chi^t \chi^a$$
$$\chi^a = \alpha \left(\hat{t}^a + \Omega \hat{\phi}^a \right)$$

$$\bar{\chi}^a = \hat{t}^a + \Omega \hat{\phi}^a$$

For $\bar{a} \ll 1$

$$\chi^a \hat{\phi}_a = 0$$

$$g_{\phi t} + g_{\phi\phi} \Omega = 0$$

ZAMO:

$$\Omega = \frac{d\phi}{dt} = - \frac{g_{t\phi}}{g_{\phi\phi}}$$

$$\bar{a} = 1$$

v^s

$$d\phi > 0$$

$$f^a + \Omega \phi^a$$

$$g_{\phi t} + g_{\phi \phi} \Omega = 0$$

$$\Omega = \frac{d\phi}{dt} = -\frac{g_{\phi t}}{g_{\phi \phi}}$$

As $r \rightarrow r_+$

$$\Omega = \frac{\bar{a}}{2r_+} = \Omega_H$$

$$\bar{K}^a = f^a + \Omega_H \phi^a$$

For $\bar{a} \ll 1$

$$r_+ \approx (2 - \frac{1}{2} \bar{a}^2) M$$

$$\bar{a} = 1$$

$$\Omega_H \approx \frac{\bar{a}}{4M}$$

$$\Omega_H = \frac{1}{2M}$$

$$\frac{dt}{d\tau}$$

Constants:

$$\tilde{E} = -\dot{t} = -U_t = -\left(g_{tt}U^t + g_{t\phi}U^\phi\right)$$

$$= \left(1 - \frac{2M}{r}\right) \frac{dt}{d\tau} + \frac{2Ma \sin^2 \theta}{r} \frac{d\phi}{d\tau}$$

$$\tilde{J} = \dot{\phi} = U_\phi = g_{\phi\phi}U^\phi + g_{t\phi}U^t$$

$$= \frac{(r^2 + a^2) \sin^2 \theta}{r} \frac{d\phi}{d\tau} - \frac{2Ma \sin^2 \theta}{r}$$

$\tilde{E} < 1$
bound orbits

Geodesic Eqn: $U^a \nabla_a U^b = 0$, $U^a = \frac{dx^a}{d\tau}$

$$U^a U_a = -1$$

$$\frac{dx^a}{d\tau} + \Gamma_{bc}^a \frac{dx^b}{d\tau} \frac{dx^c}{d\tau} = 0$$

$$U^b \nabla_b (K_a U^a) = \frac{1}{2} U^a U^b \nabla_b K_a = 0$$

$E < 1$
bound orbits

Consta

$\tilde{E} =$

$\tilde{J} =$

Carter Constant

$$C = k_{ab} U^a U^b$$

Restrict $\theta = \frac{\pi}{2}$, $U^\theta = 0 = U_\theta$
Equatorial geodesics

$$(*) = -$$

$$= \frac{(r+a)^2 - \Delta a \sin^2 \theta}{\Sigma} \sin^2 \theta \frac{d\phi}{dt}$$

$$(*) = -1 = g^{ab} u_a u_b = g^{tt} \tilde{E}^2 + g^{rr} (u_r)^2 + g^{\phi\phi} (\tilde{J})^2 - 2g^{t\phi} \tilde{E} \tilde{J}$$

$$g^{tt} = \frac{-((r^2+a^2)^2 - a^2 \Delta)}{\Delta r^2}, \quad g^{rr} = \frac{\Delta}{r^2}, \quad g^{\phi\phi} = \frac{\Delta - a^2}{\Delta r^2}$$

$$-1 = \frac{r^2}{\Delta} (u^r)^2 + \frac{a^2 \Delta - (r^2+a^2)^2}{\Delta r^2} \tilde{E}^2 + \frac{(\Delta - a^2)}{\Delta r^2} \tilde{J}^2 + \frac{4Ma}{\Delta r} \tilde{E} \tilde{J}$$

$$-(u^r)^2 = (1 - \tilde{E}^2) - 2M\left(\frac{1}{r}\right) + [a^2 (1 - \tilde{E}^2) + \tilde{J}^2] \left(\frac{1}{r^2}\right) - 2M(a\tilde{E} - \tilde{J})^2 \left(\frac{1}{r}\right)$$

$$= 2V(\tilde{E}, \tilde{J}, r)$$

$$= \frac{(1+a) \sin^2 \theta \frac{d\theta}{dr}}{\Sigma} - \frac{\frac{d\theta}{dr}}{\Sigma}$$

$$g^{ab} u_a u_b = g^{tt} \tilde{E}^2 + g^{rr} (v_r)^2 + g^{\phi\phi} (\tilde{J})^2 - 2g^{t\phi} \tilde{E} \tilde{J}$$

$$g^{tt} = \frac{-((r^2+a^2)^2 - a^2 \Delta)}{\Delta r^2}, \quad g^{rr} = \frac{\Delta}{r^2}, \quad g^{\phi\phi} = \frac{\Delta - a^2}{\Delta r^2}, \quad g^{t\phi} = \frac{-2Ma}{\Delta r}$$

$$= \frac{a^2 \Delta - (r^2+a^2)^2 \tilde{E}^2}{\Delta r^2} + \frac{(\Delta - a^2)}{\Delta r^2} \tilde{J}^2 + \frac{4Ma}{\Delta r} \tilde{E} \tilde{J} - 2M \left(\frac{1}{r}\right) + [a^2 (1 - \tilde{E}^2) + \tilde{J}^2] \left(\frac{1}{r^2}\right) - 2M(a \tilde{E} - \tilde{J})^2 \left(\frac{1}{r^2}\right)$$

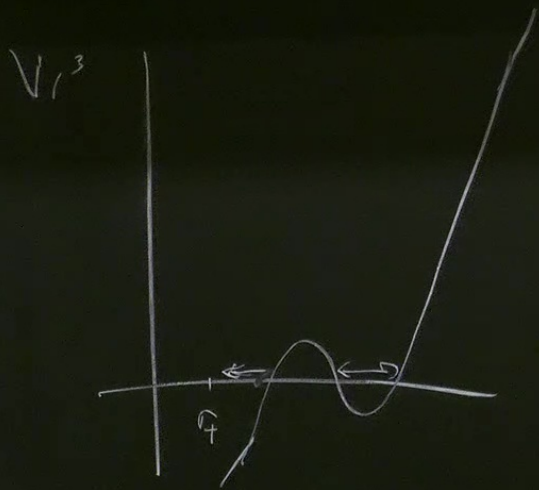
r)

Need $V \leq 0$, turning points when $V=0$ ($\Rightarrow \frac{dr}{d\tau}=0$)

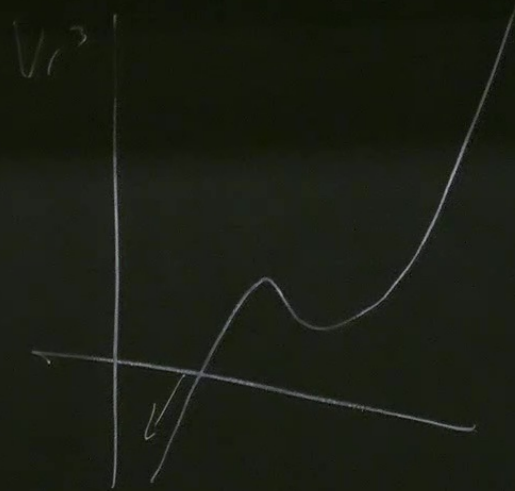
$\tilde{E} < 1$ for $r \rightarrow \infty$ $V(r) \rightarrow \frac{1}{2}(1-\tilde{E}^2) > 0$

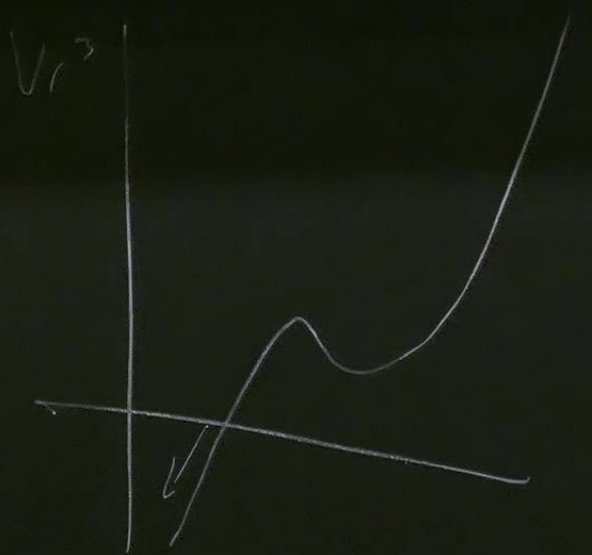
For $r=r_+$

$$\begin{aligned} 2r_+^4 V(r_+) &= -(2Mr_+ \tilde{E})^2 - (a\tilde{J})^2 + 4Mar_+ \tilde{E}\tilde{J} \\ &= -(2Mr_+ \tilde{E} - a\tilde{J})^2 \leq 0 \end{aligned}$$



δr



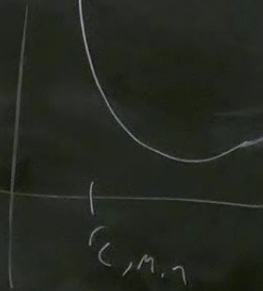


Marginal case, $V(r_c) = \frac{dV}{dr}(r_c) = 0$
for $r_c > r_+$

$$E_c = \frac{1 - \frac{2M}{r_c} + \frac{a}{r_c} \sqrt{\frac{M}{r_c}}}{\sqrt{1 - \frac{3M}{r_c} + \frac{2a}{r_c} \sqrt{\frac{M}{r_c}}}}$$

$$J_c = \frac{\sqrt{Mr_c} - 2a \frac{M}{r_c} + \frac{a^2}{r_c} \sqrt{\frac{M}{r_c}}}{\sqrt{1 - \frac{3M}{r_c} + \frac{2a}{r_c} \sqrt{\frac{M}{r_c}}}}$$

For $r_c \gg M$



L_1

u

Step

See

$$= \frac{(r^2 + a^2)^2 - \Delta a^2 \sin^2 \theta}{\Sigma} \sin^2 \theta \frac{dt}{dr} - \frac{2Mar \sin^2 \theta}{\Sigma} \frac{dt}{dr}$$

$$1 - \frac{M}{2rc}$$

$$\sqrt{Mr_c}$$

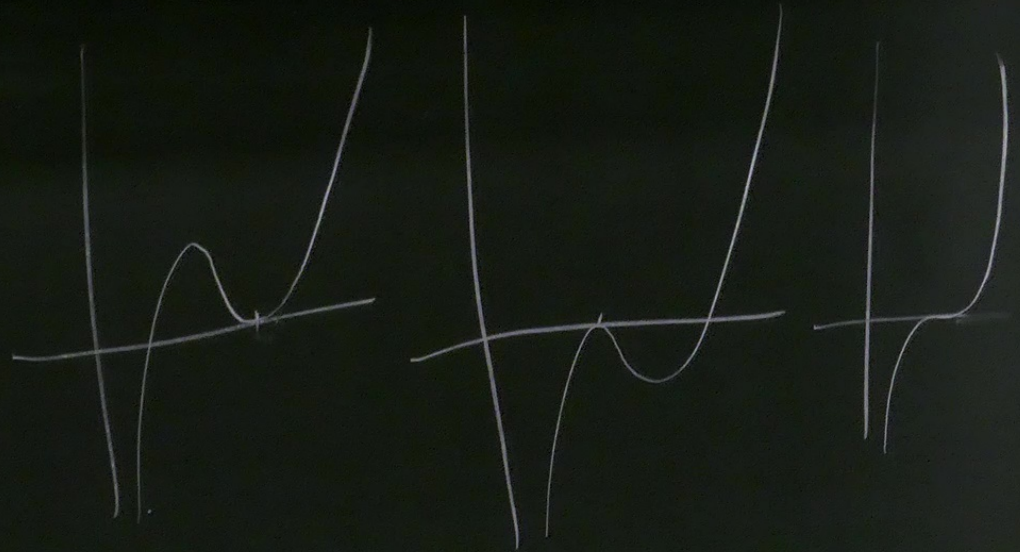
$$r_{c, \min} = 2M \left[1 + \cos \left(\frac{2}{3} \cos^{-1}(\bar{a}) \right) \right]$$

$$\bar{a} = 0, \quad r_{c, \min} = 3M$$

$$\bar{a} = 1, \quad r_{c, \min} = M$$

$$\bar{a} = -1, \quad r_{c, \min} = 4M$$

Marginal case, $V(r_c) = \frac{dV}{dr}(r_c) = 0$
for $r_c > r_+$
 $\frac{d^2V}{dr^2} = 0$



Σ $d\tau$ 2 $d\tau$

$$r_{c, \min} = 2M \left[1 + \cos \left(\frac{2}{3} \cos^{-1}(\bar{a}) \right) \right]$$

$$\bar{a} = 0, \quad r_{c, \min} = 3M$$

$$\bar{a} = 1, \quad r_{c, \min} = M$$

$$\bar{a} = -1, \quad r_{c, \min} = 4M$$

$$r_{\text{ISCO}}^2 - 6Mr_{\text{ISCO}} + 8a\sqrt{M} r_{\text{ISCO}}^{1/2} - 3a^2 = 0$$

$$a = 0, \quad r_{\text{ISCO}} = 6M$$

$$a = M, \quad r_{\text{ISCO}} = M$$

$$a = -M, \quad r_{\text{ISCO}} = 9M$$

$$-\frac{M}{2r_c}$$

$$M r_c$$

$$(r_{\text{ISCO}}) = \text{circle}$$

 V_{r^3}

prograde / retro $d\ell^2$

$$r_{\text{ISCO}}/M = 3 + B \mp \sqrt{(3-A)(3+A+2B)}$$

$$A = 1 + (1 - \bar{a}^2)^{1/3} \left[(1 + \bar{a})^{1/3} + (1 - \bar{a})^{1/3} \right]$$

$$B = \sqrt{3\bar{a}^2 + A^2}$$

$r_{\text{c,min}} < r < r_{\text{ISCO}}$
 unstable

